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Effects of ambient pressure on ignition and flame characteristics in diesel spray combustion

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Abstract

This work reports on numerical investigation of effects of ambient pressure (P_{am}) on spray combustion under engine-like conditions. Three cases with different P_{am} of 42, 85 and 170 bar at a fixed ambient temperature of 1000 K are considered. Zero-dimensional calculations are first performed for autoignition of stagnant adiabatic homogenous mixtures to evaluate performance of the selected diesel surrogate fuel models and to identify the P_{am} effects on the most reactive mixture. An Eulerian-based transported probability density function model is then chosen for the three-dimensional computational fluid dynamics study. The results show the predicted ignition delay times and flame lift-off lengths are in reasonably good agreement with experiment, with the relative difference below 28%. The current work reveals that low-temperature reactions occur across a wide range of mixture fraction but a noticeable rise of temperature (>100 K above ambient temperature) is detected first on the fuel-lean side of the stoichiometric line in all three cases. The high-temperature ignition occurs first on the fuel-rich side in the 42 and 85 bar cases, where the igniting mixture appears to be more fuel-rich in the latter case. As P_{am} is further

increased to 170 bar, the igniting mixture becomes more fuel-lean and the high-temperature ignition occurs on the fuel-lean side. The ignition behavior is found to depend on both physical and chemical processes. At 170 bar, the reaction rate increases and the associated transition from low- to high-temperature ignition is relatively fast, as compared to the transport of warmer products from the lean zone into the fuel-rich mixture. Also, within the fuel-rich region, the local temperature is low due to liquid fuel vaporization and the condition is not appropriate for ignition. These collectively cause the high-temperature ignition to occur on the fuel-lean side. Analyses on the quasi-steady spray flame structures reveal that, apart from poorer air entrainment due to reduced lift-off length, the higher rich-zone temperature and lower scalar dissipation rate also lead to a higher peak soot volume fraction at higher P_{am} .

Keywords: spray flame, transported probability density function, ignition, pressure effects

1. Introduction

Over the last few decades, automotive, light-duty, heavy-duty and marine engine industries have been working on improving the understanding of in-cylinder phenomena, which is an important prerequisite to design clean and efficient engines [1-12]. Numerous research groups, mainly led by the Sandia National Laboratory, share their optical measurements and modelling results through the Engine Combustion Network (ECN) [3], with the aim to facilitate and consolidate diesel spray combustion research. Effects of a wide range of diesel engine parameters on fuel penetration, ignition delay time (IDT), flame lift-off length (LOL) and soot volume fraction ($f_{v,soot}$) have been studied. However, the majority of these studies were performed for an ambient density (ρ_{am}) range of 14.8 to 22.8 kg/m³. In particular, the Spray A condition, where ρ_{am} is fixed at 22.8 kg/m³ has been broadly studied in the last couple of years, with an emphasis on understanding the ignition and flame stabilization processes [4-7]. The initial ambient air temperature before the air is cooled by the vaporized fuel, T_{am} , is set to 900 K, corresponding to

an ambient pressure (P_{am}) of ~67 bar. Under the typical range of T_{am} at the start of injection in heavy-duty and large two-stroke marine engines, the ambient densities can be approximately 30.0 and 60.0 kg/m³ under part and full load conditions, respectively [10,12-15]. Optical accessible experiments conducted at such a high ρ_{am} (or P_{am}) were reported by Siebers and co-workers [8-12]. In their experimental studies, the ρ_{am} was varied from 7.3 to 58.5 kg/m³, corresponding to a P_{am} of ~20 to ~170 bar at a T_{am} of 1000 K. Under these test conditions, both the liquid and vapor penetration lengths decreased with increasing P_{am} . Besides that, IDT and flame LOL were found to decrease with increasing P_{am} , while the spray flame size at quasi-steady state (in terms of both length and width) reduced when P_{am} increased [10,11]. The effects of P_{am} on ignition and premixed combustion processes up to 45.0 kg/m³ (~120 bar at T_{am} of 1000 K) were also investigated by Higgins et al. [12]. In terms of soot formation, measurements from optical accessible experiments showed path-length-averaged $f_{v,soot}$ and maximum $f_{v,soot}$ increased with P_{am} [3,13]. Nevertheless, it is challenging to deduce the low- and high-temperature ignition as well as other combustion phenomena solely based on experimental observation at such elevated P_{am} . To date, the effects of P_{am} on spray flame structures under these conditions were not explicitly studied numerically, although these experimental data have been used for computational fluid dynamic (CFD) model validation [16,17]. A more comprehensive understanding of P_{am} effects is crucial for both modelling of engine combustion and for developing physical and chemical models at these P_{am} levels.

Set against this background, the objectives of this CFD work are first to generate velocity, species and temperature distribution profiles to help improving the understanding of the ignition/combustion process at high P_{am} levels; and second to elucidate the effects of P_{am} on spray flame structure and emissions formation. These aims are achieved by employing an Eulerian-based transported probability density function (PDF) method [18] and a skeletal n -heptane mechanism [19]. Zero-dimensional calculations are also performed for autoignition of stagnant adiabatic homogenous mixtures to complement the CFD study.

The remainder of the paper is structured as follows. In the next section, experimental data used for model validation are first described. It is followed by the descriptions of the numerical methods. The subsequent sections detail the model validation as well as the numerical analyses of autoignition and flame structures at different P_{am} levels. Conclusions from this work are outlined in the final section.

2. Experimental data for model validation

The grade number two diesel fuel (diesel #2) spray experimental data used for model validation in this work were obtained in a constant volume pre-burn chamber [9,10]. The initial ambient gas composition is determined by the composition of the combustible-gas mixture burned to generate diesel engine-like temperatures and pressures. In the selected test conditions, the reaction $3.245 \cdot \text{C}_2\text{H}_2 + 0.515 \cdot \text{H}_2 + 8.37 \cdot \text{O}_2 + 89.75 \cdot \text{N}_2 \rightarrow 89.75 \text{ N}_2 + 6.49 \cdot \text{CO}_2 + 3.76 \cdot \text{H}_2\text{O}$ was used to generate the inert high-pressure, high-temperature environment in the non-reacting spray cases. A combustible-gas mixture of 68.1% N_2 , 28.4% O_2 , 3.0% C_2H_2 and 0.5% H_2 (by volume) was then used in the reacting spray cases. The product composition of this combustible mixture simulated air, having a composition of 21.0% O_2 , 69.3% N_2 , 6.1% CO_2 , and 3.6% H_2O (by volume) [10-12]. The experimental initial ambient gas composition is used in the current reacting spray cases in order to produce identical initial thermochemical conditions. The model performance in simulating fuel penetration lengths of evaporating, non-reacting sprays (cases 1 to 3 shown in Table 1) at different P_{am} is first evaluated. For liquid-phase, the comparison is made against the liquid penetration length determined with the liquid length scaling law. The properties of *n*-heptadecane are used in the liquid length scaling law to produce diesel #2 liquid length [8] since these resemble the properties of diesel #2. The simulated vapor penetration length is compared with the measurements reported by Naber and Siebers [9]. For the validation of the reacting sprays (cases 4 to 6 shown in Table 1), IDT and LOL measurements from the ECN database [3] are used; more descriptions can be found in Ref. [10].

Table 1. Operating conditions and injection specifications

Case	$O_{2,am}$	T_{am} [K]	ρ_{am} (kg/m ³)	D_{nozz} (μ m)	P_{inj} (bar)	\dot{m}_f (mg/ms)
1	0	1000	13.9	257	1400	14.0
2	0	1000	28.6	257	1400	14.0
3	0	1000	58.6	257	1400	14.0
4*	21%	1000	14.8	180	1400	8.8
5*	21%	1000	30.0	180	1400	9.0
6*	21%	1000	58.5	180	1400	9.2

*Note: Non-reacting, vaporising spray simulations are also carried out for cases 4 to 6, where $O_{2,am}$ is set to 0.

As depicted in Table 1, three cases with different ρ_{am} of 14.8, 30.0 and 58.5 kg/m³ are considered for the reacting conditions. At a fixed T_{am} of 1000 K, these correspond to P_{am} of 42, 85 and 170 bar, respectively. Initial species composition, flow and turbulence conditions can be found in a previous work [20]. The operating conditions as well as the injection specifications including nozzle diameter (D_{nozz}), injection pressure (P_{inj}) and fuel mass flow rate (\dot{m}_f) of the test cases are listed in Table 1.

3. Numerical methods

3.1 Autoignition of homogeneous mixtures

To date, effects of P_{am} on autoignition of various hydrocarbon fuels have been studied in shock tube and plug flow reactor experiments as well as kinetic model simulations. Hashemi *et al.* [21-23] recently studied the effects of P_{am} on C₁-C₃ hydrocarbon reacting pathway up to 100 bar. However, for larger hydrocarbons which are commonly used as diesel #2 surrogate fuels such as *n*-heptane [24,26], *n*-dodecane/myxylene [27] and diesel primary reference fuels [28], the tested P_{am} only considered up to ~50 bar. To the authors' knowledge, validation of diesel #2 surrogate fuel kinetic models for pressure levels of 80 bar and above (targeted conditions in the current test cases) is not available. Thus, five different chemical kinetic mechanisms are evaluated at different P_{am} levels. These include the detailed Lawrence Livermore National Laboratory *n*-heptane model (the Mehl654 mechanism) [24] and two of its skeletal versions (the Liu44 and Lu68 mechanisms) [19,25], a recently updated *n*-heptane model (the Zhang1268 mechanism)

[26], and an integrated model of *n*-dodecane and myxylene (the Pei163 mechanism) [27]. To evaluate the selected chemical mechanisms, zero-dimensional (0-D) calculations are conducted for autoignition of stagnant adiabatic homogenous mixtures at three different initial P_{am} of 42, 85 and 170 bar using the CHEMKIN-PRO software [29]. The IDT is defined as the time where the mixture temperature increases to 400 K above the initial temperature ($T_{t=0}$). A quadratic relation, $T_{t=0} = T_{am} + 1120Z^2 - 1370Z$, is used to correlate the temperature and mixture fraction. Here, Z is the Bilger mixture fraction while T_{am} is the initial ambient temperature in the combustion chamber and is set to 1000 K. This relation is obtained by post-processing the 3D-CFD results of the reacting case prior to the start of low-temperature ignition, corresponding to the mixing between the cold fuel and hot air. The quadratic relation is constructed for $0 \leq Z \leq 0.4$ and the same function is found applicable to represent the mixing at all three P_{am} (*cf.* Figs. 7 to 9) such that P_{am} is the only variable in the comparison. Akin to that in Ref. [30], the most reactive mixture (Z_{mr}) is characterized as the mixture which has the shortest IDT, and the associated IDT can serve as a reference for the comparison against the results from inhomogeneous and turbulent mixtures [30]. It may be worth mentioning that effects of pressure on Z_{mr} have not yet been explicitly studied for turbulent non-premixed flames [31]. Based on the results of Fig. 1, all the models produce similar qualitative trends with respect to the change of P_{am} , *i.e.*, the Z_{mr} occurs in more fuel-lean mixture and Z_{mr} falls closer to the stoichiometric mixture fraction, Z_{st} when P_{am} increases (*cf.* Fig. 1). The results also show that the IDT predicted by the Liu44 mechanism is close to those of its detailed counterpart, in particular near the Z_{mr} regimes. This trend is similar to that reported by Pei et al. [32] under a similar ambient, thermochemical condition, although the fuel temperature is lower at 373 K in their cases.

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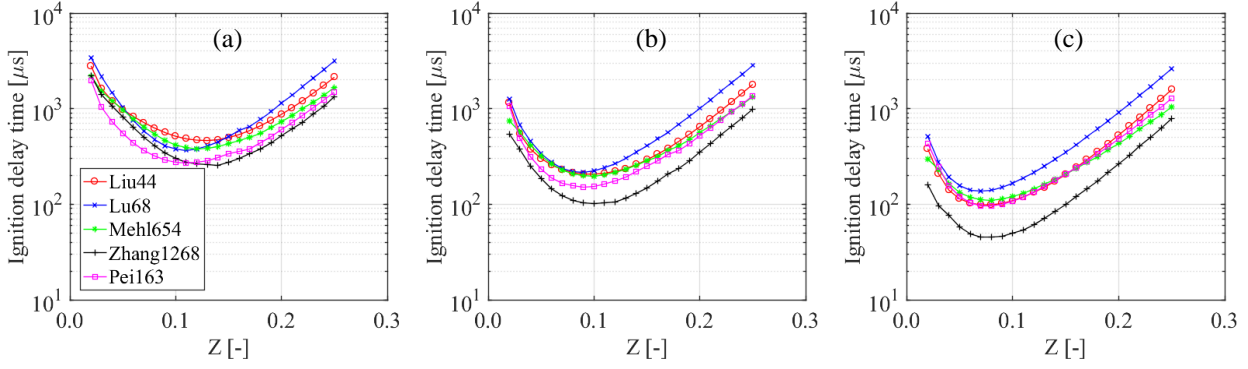


Fig. 1. Ignition delay times of homogenous mixtures for varying initial mixture fraction and temperatures at (a) 42 bar, (b) 85 bar and (c) 170 bar.

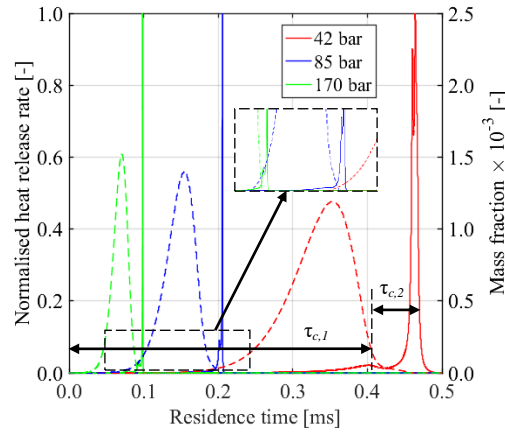


Fig. 2. Ignition histories of Z_{mr} computed in the zero-dimensional homogenous reactor model using the Liu44 mechanism. Solid lines denote the normalized heat release rate while dotted lines represent the mass fraction of RO_2 .

In the case where the Liu44 model is used, Z_{mr} are recorded to be 0.13, 0.10 and 0.07 in 42, 85 and 170 bar case, respectively. The corresponding $T_{t=0}$ are 841 K, 874 K and 910 K, respectively. The simulated heat-release rate (HRR) profiles are plotted in Fig. 2. Temporal evolution of the heptyl radical (RO_2) of these cases are also provided to indicate the low-temperature ignition. Here, the IDTs coincide at the peaks of the HRR. The IDT can be expressed as $IDT = \tau_{c,1} + \tau_{c,2}$, where $\tau_{c,1}$ and $\tau_{c,2}$ are the first-stage (low-temperature) and second-stage (high-temperature) IDTs respectively. All the mixtures in these cases undergo two-stage ignition, where

the high-temperature ignition occurs after the high amount of RO_2 is consumed. It is also noted in the insets of Fig. 2 that $\tau_{c,2}$ becomes negligible when the P_{am} increases. The Liu44 mechanism is therefore chosen for the 3-D CFD simulations, which are described next.

171

3.2 CFD model formulation

The 3-D CFD spray combustion simulations are carried out using OpenFOAM version 3.0.1 [33]. Detailed descriptions of the model can be found in [35] and only a brief description is provided here. The fuel spray, flow and combustion processes are modelled using the Eulerian-Lagrangian approach within the unsteady Reynolds averaged Navier-Stokes (URANS) framework. The Reitz-Diwakar model is used to simulate the fuel droplet breakup, while the Realizable $k-\varepsilon$ model is employed for turbulence modelling. For the nitric oxide (NO) formation modelling, the Zeldovich thermal NO sub-mechanism from Ref. [34] are added to the Liu44 mechanism. The interaction between turbulence and chemistry is modelled using the Eulerian-based transported PDF method, Eulerian Stochastic Field (ESF) [18]. Similar to the Lagrangian particle transported PDF method, the ESF method is a general approach to account for the turbulence-chemistry interaction (TCI). It can be used to simulate the autoignition as well as different levels of ‘premixedness’ *i.e.*, premixed, partially premixed and non-premixed. In the ESF method, the governing equation for the n -th stochastic field is

$$\begin{aligned} \overline{\rho} d\phi_a^{(n)} = & -\overline{\rho} \tilde{u}_i \frac{\partial \phi_a^{(n)}}{\partial x_i} dt + \overline{\rho} S_{\alpha}^r(\phi^{(n)}) dt + \overline{\rho} S_{\alpha}^s(\phi^{(n)}) dt \\ & + \frac{\partial}{\partial x_i} \left(\Gamma_t \frac{\partial \phi_a^{(n)}}{\partial x_i} \right) dt - \frac{1}{2} \overline{\rho} C_{\phi} (\phi_a^{(n)} - \tilde{\phi}_a) \omega_i dt + \overline{\rho} \sqrt{2 \frac{\Gamma_t}{\rho}} \left(\frac{\partial \phi_a^{(n)}}{\partial x_i} \right) dW_i^{(n)} \end{aligned} \quad (1)$$

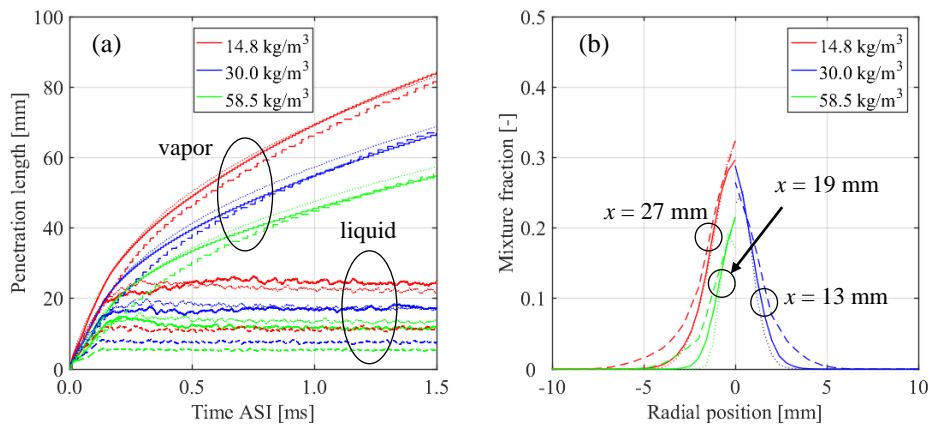
where ϕ_a denotes the mass fraction of species (Y_i) or the enthalpy of the mixture (h), and $\phi^{(n)} = [Y_1^{(n)}, Y_2^{(n)}, \dots, Y_i^{(n)}, h^{(n)}]$. The first term on the right hand side (r.h.s.) represents the convective term while the second term, $\overline{\rho} S_{\alpha}^r(\phi^{(n)}) dt$, is due to the chemical reactions. The third term, $\overline{\rho} S_{\alpha}^s(\phi^{(n)}) dt$, is the source term due to the spray evaporation. This is different for each species, *i.e.*, solely the evaporating species is involved but it is identical for each stochastic field.

192 The fourth term corresponds to a gradient transport model for turbulent velocity fluctuation,
 193 where $\Gamma_t = \mu_t / \sigma_t$ is the turbulent diffusivity. Here, μ_t is the turbulent viscosity while σ_t is the
 194 turbulent Schmidt number (Sc_t) in the transport equations for chemical species or the Prandtl
 195 number (Pr_t) in the enthalpy transport equation. Both Sc_t and Pr_t are set to 0.7 in the current
 196 work. The fifth term, which involves the mixing constant, C_ϕ , represents the molecular mixing.
 197 The mixing constant is fixed at $C_\phi = 2$. It is modelled using the Interaction with Exchange to the
 198 Mean model. ω_t therein is the turbulence frequency obtained from $\omega_t = \varepsilon / k$ where k and ε are
 199 the turbulent kinetic energy and its dissipation rate, respectively. In the last term on the r.h.s.,
 200 $dW^{(n)}$ represents a vector Wiener process that is spatially uniform but different for each field.
 201 The purpose of this term is to introduce a stochastic noise in the transport equations and generate
 202 consequent PDF for chemical species and enthalpy. In these simulations, $dW^{(n)}$ is represented by
 203 a time-step increment $\Delta t^{1/2} \eta_i^n$, where $\eta_i^n = \{-1, 1\}$ is a dichotomic random vector. Previous work
 204 has shown that the use of thirty-two stochastic fields reached result convergence [35]. The
 205 Chemistry Coordinate Mapping (CCM) approach is coupled with the ESF solver in order to
 206 integrate the chemical reaction source terms efficiently [36]. In the current work, a four-
 207 dimensional phase space based on temperature, local equivalence ratio, scalar dissipation rate,
 208 and the mass fraction of fuel is used. Their resolutions are fixed at 5 K, 0.01, 0.025 and 0.001
 209 respectively. The ESF-CCM solver is incorporated with a revised multi-step soot model [14,20].
 210 Mean molar concentrations of acetylene (C_2H_2) and O_2/OH are used to estimate the soot
 211 formation and oxidation rates respectively. Both soot and gas radiation heat transfers are not
 212 considered in the current work. Bolla et al. [37] showed that the effect of radiation heat transfer
 213 influences the local flame temperature in the order of O(10 K) and hence has minor influence on
 214 local NO and soot concentrations under the Spray A condition. While the effects of radiation heat
 215 transfer are more significant at higher P_{am} due to the increased concentrations of soot, CO_2 and
 216 H_2O , they may only vary the absolute values. Comparisons to the experimental results in section

217 4 show variation of combustion and emissions characteristics with respect to the change of P_{am}
 218 levels are captured by the current model. These qualitative trends are expected to remain
 219 unchanged with the consideration of radiation heat transfer.

220 The computational domain is a constant volume chamber, which has a cubic shape with
 221 side lengths of 108 mm. The injector is placed at the center of one of the chamber walls. A
 222 uniform Cartesian grid is used. In the previous study, an isotropic cell size of 0.5 mm within the
 223 spray combustion region was found to reach mesh independence [35]. This mesh resolution is set
 224 as the baseline configuration and mesh sensitivity studies are carried out for different P_{am} in the
 225 current work. The computational grid with the fine, intermediate (baseline) and coarse resolutions
 226 consists of approximately 1,440,000, 360,000 and 89,000 cells, respectively. Figure 3 shows that
 227 the penetration lengths and mixture fraction generated by the finer resolution and the baseline are
 228 close while those by the coarser resolution deviate, particularly for liquid fuel penetration length.
 229 It should also be mentioned that the qualitative change of the ignition behavior (see Section 4)
 230 with respect to the variation of P_{am} is also consistent when a finer resolution is used.

231



232

233 Fig 3.(a) Penetration lengths and (b) mixture fraction distribution at axial position (x) near the
 234 end of liquid penetration length predicted using different mesh resolution. Dashed-, solid- and
 235 dotted-lines are results of the coarse, intermediate (baseline) and fine resolutions, respectively.

236

237

4. Results and discussion

4.1 Model validation

Figure 4 depicts the temporal evolution of the liquid and vapor penetration lengths of the evaporating, non-reacting sprays from the experiments and the numerical models. The liquid penetration reaches steady state after 0.5 ms, while the vapor penetration continues to grow with time. The model is capable to replicate the experimental penetration lengths reasonably well. As the P_{am} increases and hence the ρ_{am} increases, both the liquid and vapor penetration lengths decrease due to the increased droplet drag caused by the higher ρ_{am} .

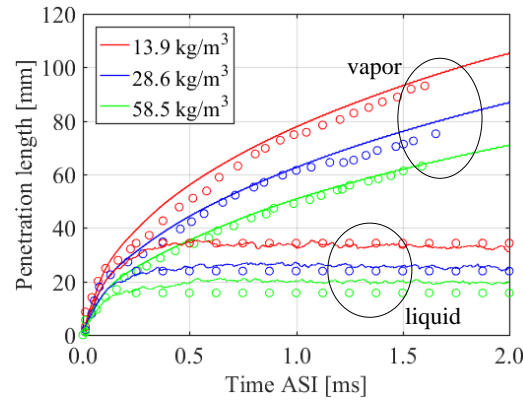


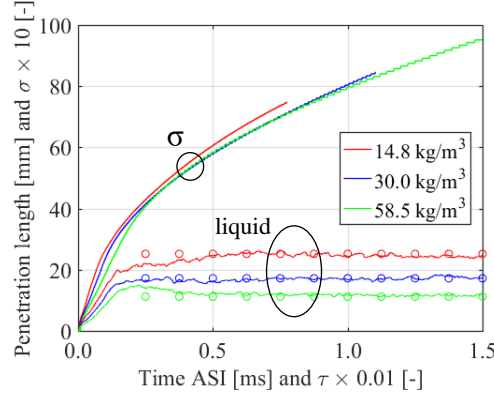
Fig. 4. Liquid and vapor penetration lengths of the evaporating, non-reacting sprays (cases 1 to 3) as a function of time after start of injection (ASI). Solid lines denote the simulation results while markers are obtained by liquid scaling law (liquid) [8] and from the experiment (vapor) [9].

Model evaluation is performed further for evaporating, non-reacting sprays with D_{nozz} of 180 μm . Since experimental data are not available for D_{nozz} of 180 μm , model evaluation is carried out based on liquid scaling law [8] as well as dimensionless penetration length and time [9]. The current spray breakup model configuration is found applicable for both D_{nozz} of 180 and 257 μm , *i.e.*, the same spray sub-model constants were used. Figure 5 shows that the predicted liquid penetration length of the non-reacting sprays for D_{nozz} of 180 μm (cases 4 to 6 with $O_{2,am} = 0\%$) agree with the liquid scaling law [4]. The associated vapor penetration length is evaluated using the dimensionless spray tip penetration distance (σ) and dimensionless penetration time (τ).

260 As seen in Fig. 5, all the σ curves collapse, agreeing well with the trend reported in [5]. This
 261 indicates that the dependence of vapor penetration length on ρ_{am} is successfully simulated by the
 262 model.

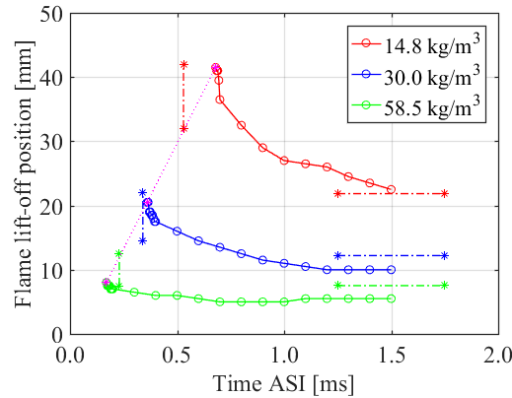
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264



265 Fig. 5. Predictions of liquid penetration lengths and dimensionless penetration distance (σ) as a
 266 function of dimensionless penetration time (τ). Circle markers denote liquid scaling law results.

267



268

269 Fig. 6. Flame lift-off positions of the reacting sprays (cases 4 to 6) as a function of time after start
 270 of injection (ASI). Vertical and horizontal dash-dotted lines with asterisks represent the
 271 experimental ignition delay times and averaged flame liftoff lengths respectively [3,10].
 272 Magenta dotted line indicates the boundary separating the domain of ignition (larger time ASI)
 273 and induction to ignition (smaller time ASI).

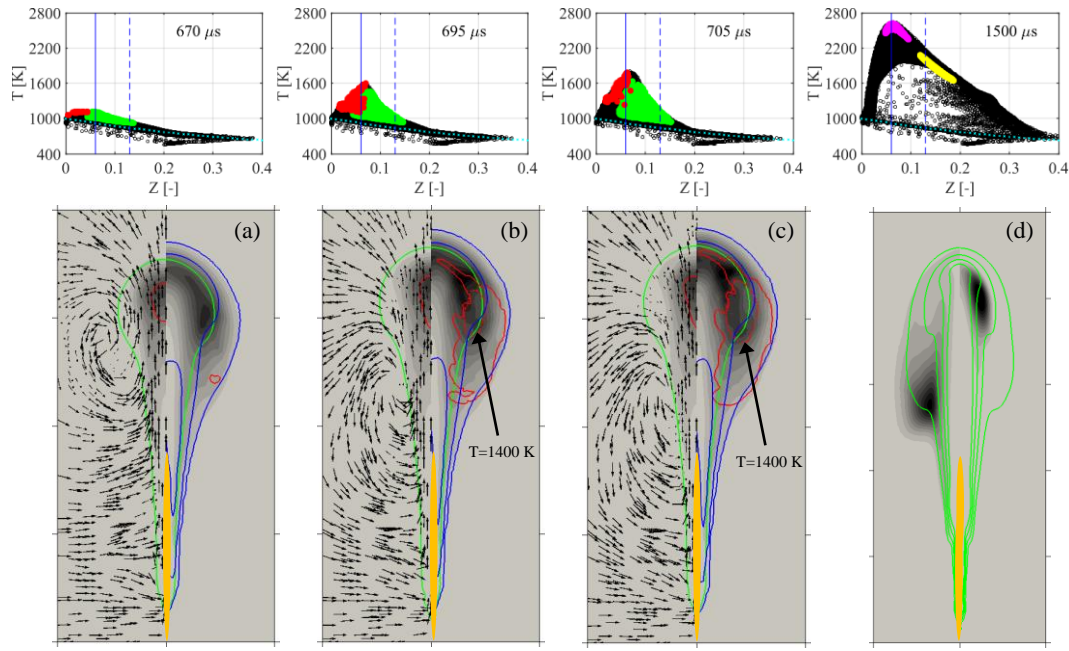
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275 The decrease in both liquid and vapor penetration lengths lead to the ignition site and the
 276 flame stabilization position to move upstream towards the injection tip, reducing the LOL (see

Fig. 6). A comparison to the measurements shows that the IDTs and flame LOLs are predicted within 28% for all three cases. The difference is mainly attributed to the overestimated IDT in the 42 bar case. It is interesting to note that a similar simulation result was reported by Bolla et al. [17], when a different TCI closure approach, the conditional moment closure model, was employed with the same mechanism for the same thermochemical conditions. Although the IDTs and LOLs predicted are to a certain extent different from the experiment, the trend of IDT and LOL with respect to the change of P_{am} is correctly predicted.

4.2 Autoignition characteristics of diesel spray combustion

Figures 7 to 9 illustrate the autoignition characteristics of the three cases using scatter plots of temperature (T)-mixture fraction (Z) and contours of various combustion products. In the 42 bar case, RO_2 is formed in the fuel-rich inner core of the spray jet with a temperature around 750 K, starting at approximately 350 μ s (not shown). Through the low-temperature reaction pathway, RO_2 oxidizes to ketohydroperoxide. The HO_2 radical is formed in the vicinity of ketohydroperoxide, extending to the fuel-lean side of the stoichiometric line. These species are formed downstream of the liquid fuel. The first-stage ignition is found across a wide range of mixture fraction. As shown in Fig. 7(a), a small amount of OH is formed at 670 μ s on the fuel-lean side and the peak temperature increases above 1100 K. Figure 7(b) depicts that HO_2 reaches a higher concentration in the fuel-rich zone at 695 μ s, as compared to that in the fuel-lean zone. HO_2 in this region is first consumed and OH is formed. The local temperature increases beyond the temperature threshold of 1400 K (400 K above T_{am}), indicating the onset of the main ignition. Rate of production (ROP) analysis of the 0-D simulation of Z_{mr} (in section 4.1) shows that OH is mainly formed from the sequence $HO_2+HO_2\leftrightarrow H_2O_2+O_2$ (R1), $H_2O_2+M\leftrightarrow 2OH+M$ (R2), where M is the third body species. It is also noticed that the main ignition occurs in the region where the scalar dissipation rate (χ) is low and the residence time is long (within the center of the recirculation zone).



303

304 Fig. 7. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion
 305 products (bottom row) in the 14.8 kg/m^3 case a) at first-stage ignition, b) at onset of second-stage
 306 ignition, c) towards the formation of a classical diesel spray flame and d) during quasi-steady
 307 state. Upper row: Cells with substantial formation of HO_2 , OH , NO and soot volume fraction
 308 ($f_{v,soot}$) are marked with green, red, magenta and yellow respectively. The thresholds are set to
 309 40% of their respective maximum mass fraction. Stoichiometric lines, the most reactive mixture
 310 fraction and mixing lines are represented by solid blue, dashed blue and dotted cyan lines
 311 respectively. Bottom row (a)-(c): (Left) Ketohydroperoxide contour with flowfield (black
 312 arrows), RO_2 of 1×10^{-4} (solid red); and (right) HO_2 contour with isolines of scalar dissipation
 313 rates of 1, 10 and 100 (solid blue), OH mass fraction of 1×10^{-4} (solid red), temperature of 1400
 314 K (dotted red); (d) (Left) NO and (right) $f_{v,soot}$ contours at quasi-steady state with isolines of local
 315 equivalence ratio of 1 to 4 (solid green). Maximum mass fraction of ketohydroperoxide, HO_2 , and
 316 NO is fixed at 0.03, 0.001 and 0.002 respectively while maximum soot volume fraction is set to 5
 317 ppm. Solid green lines indicate the stoichiometric line and the orange lines represent the averaged
 318 liquid length. For (a) to (c), each frame shows $30 \times 60 \text{ mm}$ while for (d), the frame shows $40 \times$
 319 100 mm .

320

321 The formation of RO_2 , ketohydroperoxide and HO_2 starts earlier at approximately $200\ \mu\text{s}$
 322 in the 85 bar case (not shown). Figure 8(a) depicts that OH formation is formed at $350\ \mu\text{s}$ in the
 323 fuel-lean region where the local temperature exceeds $1100\ \text{K}$. The second-stage ignition then
 324 takes place at $365\ \mu\text{s}$ on the fuel-rich side of the stoichiometric line, in the region where χ is low
 325 and the residence time is long. These phenomena are akin to those of 42 bar case discussed
 326 above, although the transition from low- to high-temperature ignition has a shorter duration. A
 327 shorter transition is also observed in the homogenous reactor calculation (*cf.* Fig. 2). Reaction
 328 path analysis shows that reaction R2 remains the most significant step during the main ignition
 329 event. When P_{am} increases, the separation between the end of liquid fuel and key radicals such as
 330 RO_2 , ketohydroperoxide and HO_2 reduces. The ignition site is closer to the penetration tip where
 331 the local Z value is higher. In the 42 and 85 bar cases, the mixture on the fuel-lean side first
 332 experiences a higher temperature rise, presumably due to the higher T_{am} . It then promotes the
 333 low-temperature reactions in the rich mixture due to the transport of warmer lean products into
 334 the fuel-rich zone. These phenomena are similar to those reported in Refs. [4-7], which studied
 335 the ignition behavior for the Spray A condition ($O_{2,am} = 15\%$; $T_{am} = 900\text{K}$; $P_{am} = 60\ \text{bar}$). The
 336 observation that the main ignition event occurs in more fuel-rich mixtures in the case with higher
 337 P_{am} is consistent with the findings of Higgins et al. [12] who experimentally studied the ignition
 338 behavior at different ρ_{am} in the same combustion vessel. However, the behavior changes when
 339 P_{am} is further increased. As discussed next, the current results show that the mixture fraction of
 340 the first igniting mixture during the high-temperature ignition does not vary monotonically with
 341 P_{am} and the second-stage ignition takes place in a more fuel-lean region in the 170 bar case.

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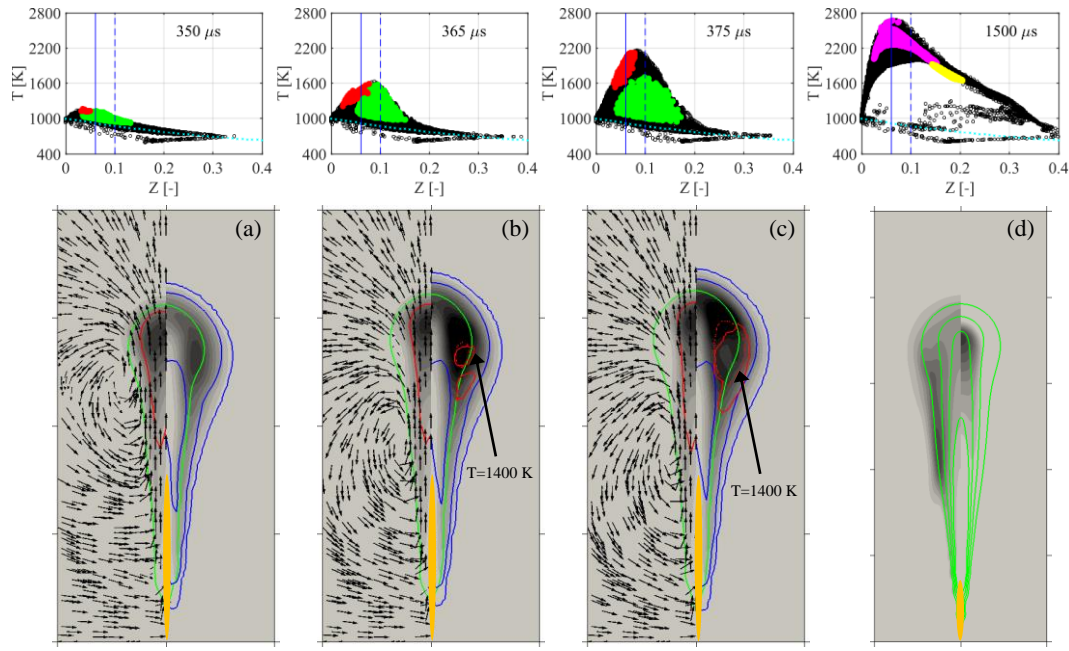


Fig. 8. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion products (bottom row) in the 30.0 kg/m³ case at different times. Descriptions can be found in the caption of Fig. 7, apart from maximum soot volume fraction which is set to 50 ppm. For (a) to (c), each frame shows 20 x 40 mm while for (d), the frame shows 40 x 100 mm.

At 170 bar, RO₂, ketohydroperoxide and HO₂ are formed early at 100 μs (not shown). The temperature of the fuel-lean mixture increases to above 1100 K at 150 μs. Under this condition, the transition from low- to high-temperature ignition is even faster, occurring in less than 15 μs. The transport of warmer products from the fuel-lean zone into the fuel-rich mixture is relatively slow, as compared to the increased reaction rate. Figure 9 depicts that there is no separation in the spray direction between HO₂ and liquid fuel, *i.e.*, HO₂ is formed in the vicinity of the fuel spray region. Within the fuel-rich region, the local temperature is low due to evaporation and the condition is not appropriate for ignition. The scatter plots in Figs. 8(b) and 9(b) support that, during the high-temperature ignition, the local temperature within the fuel-rich region becomes lower when P_{am} increases. Eventually, the high-temperature ignition occurs on the fuel-lean side of the stoichiometric line.

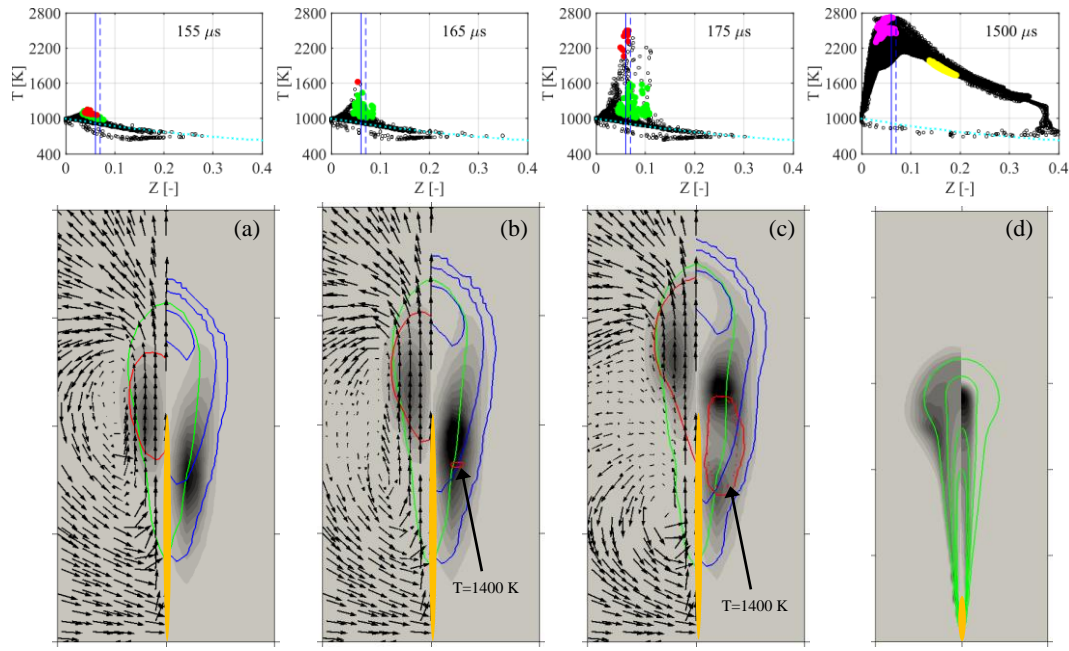
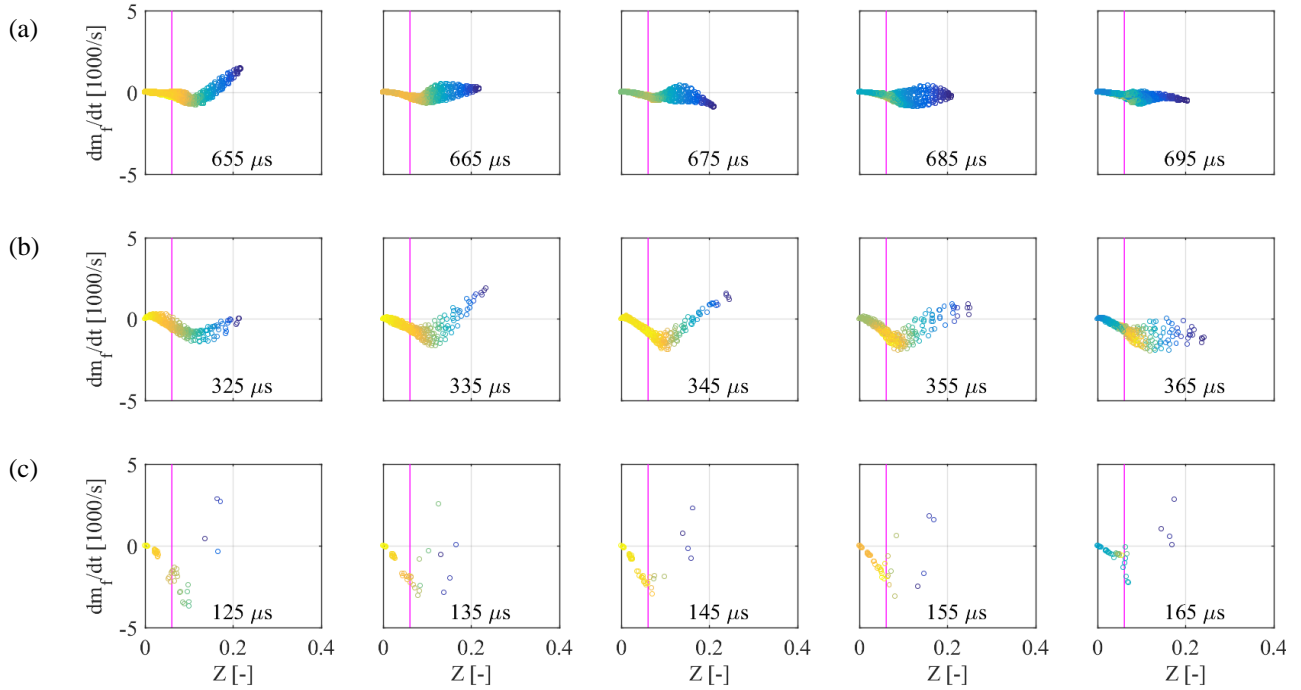


Fig. 9. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion products (bottom row) in the 58.5 kg/m^3 case at different times. Descriptions can be found in the caption of Fig. 3, apart from maximum soot volume fraction which is set to 500 ppm. For (a) to (c), each frame shows $5 \times 20 \text{ mm}$ while for (d), the frame shows $40 \times 100 \text{ mm}$.

The ignition characteristics from low- to high-temperature ignition in the three cases are also illustrated using the scatter plots of rate of fuel concentration change and mixture fraction. These plots can be found in Fig. 10. A positive value indicates the presence of vapor fuel while a negative value denotes the fuel consumption. Each plot is colored by the normalized local temperature at that particular timestep, *i.e.*, dark blue, blue, green and yellow represent low, intermediate, high, the highest temperature, respectively. At 42 and 85 bar, the higher temperature regime first falls on the fuel-lean mixture and then shifts to the fuel-rich mixture. During the main ignition process, the fuel has been consumed where the main ignition event occurs. On the other hand, at 170 bar, the vapor fuel is still observed but the corresponding mixture is cold and does not get involved in the ignition process. The mixture on the fuel-lean side of the stoichiometric line undergoes low- and then high-temperature ignition. These support the discussion earlier based on Figs 7 to 9.



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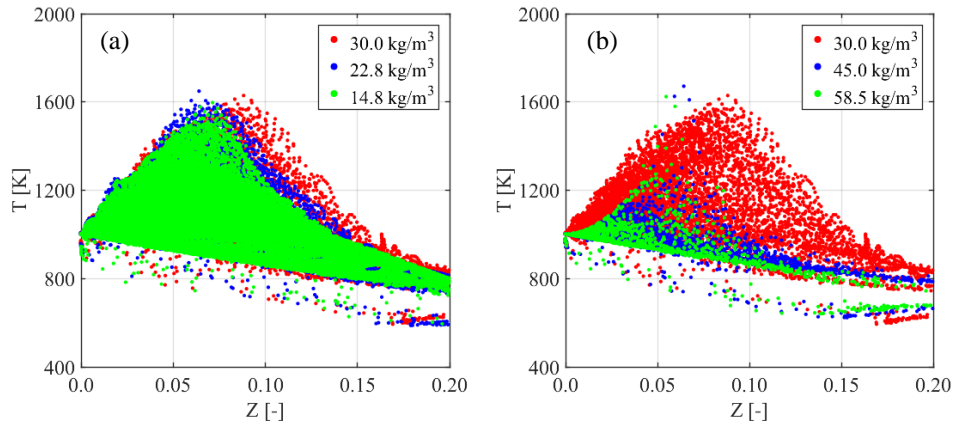
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Fig. 10. Scatter plots of rate of fuel concentration change and mixture fraction for (a) the 14.8 kg/m³ case at 44 mm from the injector tip, (b) the 30.0 kg/m³ case at 26 mm from the injector tip and (c) the 58.5 kg/m³ case at 8 mm from the injector tip. The last column provides scatter plots at their respective main high temperature ignition.



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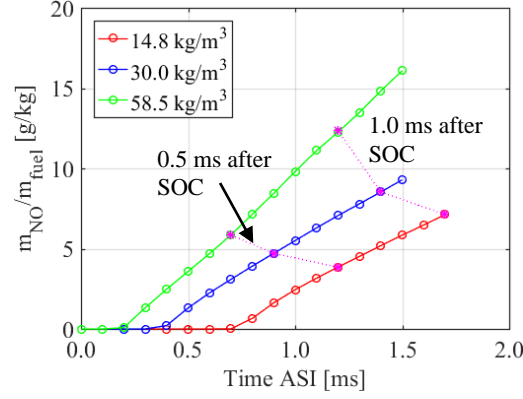
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Fig. 11. Scatter plots of temperature-mixture fraction of (a) the 14.8 kg/m³ case at 695 μs, the 22.8 kg/m³ case at 470 μs, the 30.0 kg/m³ case at 365 μs and (b) the 30.0 kg/m³ case at 365 μs, the 45.0 kg/m³ case at 230 μs and the 58.5 kg/m³ case at 165 μs.

391 Computation of two additional intermediate cases of ρ_{am} of 22.8 kg/m³ (67 bar) and 45.0
 392 kg/m³ (128 bar) confirms this trend. Figure 11 demonstrates that the main ignition occurs in more
 393 fuel-rich mixtures when P_{am} increases from 42 bar to 85 bar, consistent with results of Higgins et
 394 al. [12]. Above 85 bar, the mixture that undergoes the main ignition becomes more fuel-lean
 395 when P_{am} increases. In order to further verify this trend, simulations are performed using the
 396 Lu68 mechanism [16] for cases 5 and 6. The predicted IDT using the Lu68 mechanism is slightly
 397 closer to the measurement in the 170 bar case (case 6), with a relative difference of < 8% from
 398 the measurement, as compared to that using the Liu44 mechanism with a relative difference of
 399 ~10% from the measurement. The high-temperature ignition occurs on the fuel-rich side (Z of
 400 0.109) and the fuel-lean side (Z of 0.0603) in the 85 bar (case 5) and 170 bar (case 6)
 401 respectively, showing that the qualitative change of the ignition behavior with respect to the
 402 variation of P_{am} from 85 bar to 170 bar is the same for the Liu44 and Lu68 mechanisms. It should
 403 be noted that the variation of Z of the first ignition mixtures during the high-temperature ignition
 404 is identified for these particular injection specifications, initial ambient gas composition and T_{am} .
 405 As the ignition behavior is dependent of both physical and chemical processes, the variation of Z
 406 of the first ignition mixtures during the high-temperature ignition with respect to P_{am} may
 407 change. Understanding the ignition behavior is crucial for development of skeletal surrogate fuel
 408 mechanisms. Once the ignition behavior is identified for a narrower range of operating
 409 conditions, the size of the skeletal surrogate fuel mechanisms can be further optimized to increase
 410 the computational efficiency.

411 Effects of P_{am} on the quasi-steady state flame structures will be discussed next. Upon the
 412 ignition point, it takes approximately 160 hours for 0.1 ms on ten 64-bit Intel Ivy Bridge Xeon
 413 E5-2650 v2 8-core CPUs running at 2.60 GHz with the Liu44 mechanism. The computational
 414 cost is expected to escalate significantly with the use of the Lu68 mechanism. Hence, only the
 415 Liu44 mechanism is used for the simulations up to quasi-steady state and the associated results
 416 are used for the flame structure analyses.



418

419 Fig. 12. Normalized total mass of NO as a function of time after start of injection (ASI). Magenta
 420 dotted lines indicate the total NO mass production remains higher after their respective start of
 421 combustion (SOC).

422 An analysis based on transport budgets [38] near the lift-off position at quasi-steady state
 423 suggests the flame is stabilized by the auto-ignition process. This remains unchanged for three
 424 P_{am} levels and agrees with the findings from the literature. In addition to that, the Takeno's flame
 425 index (FI) is used to investigate the spray flame structure. The local equivalence ratio is
 426 incorporated into FI such that the premixed flame can be identified for both fuel-lean and fuel-
 427 rich conditions, as shown in Eq. (2),

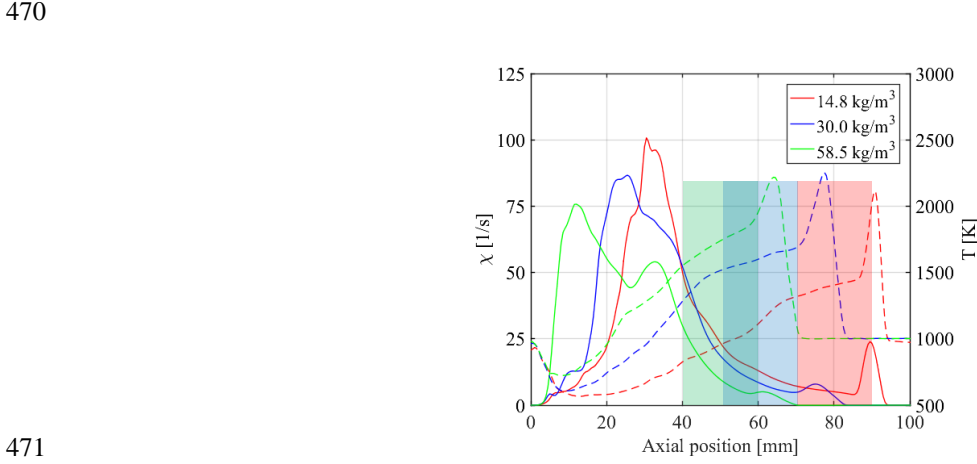
$$428 \quad FI = \frac{\nabla Y_F \cdot \nabla Y_O}{|\nabla Y_F \cdot \nabla Y_O|} \cdot \frac{\phi - 1}{|\phi - 1|} \quad (2)$$

429 where the mass fraction of fuel, Y_F includes those of evaporating species, n -heptane as well as the
 430 main fuel-rich combustion products, *i.e.* CO and H₂ [35]. The quasi-steady spray structures are
 431 found independent from the effects of P_{am} at the tested initial T_{am} and ambient gas composition.
 432 Takeno's FI shows that all the spray flames have the classical structure of the diesel spray flames
 433 described in the Dec conceptual model, *i.e.*, a premixed rich flame upstream embedded by a
 434 diffusion flame which extends downstream [40]. Figures 7(d), 8(d) and 9(d) demonstrate that the
 435 location of NO and soot formation regions for the three P_{am} levels. High NO concentration is
 436 found above 2200 K near the stoichiometric mixture, while high soot concentration is in the
 437 premixed fuel-rich zone with local temperature ranging from 1600 to 2200 K; this result is

438 consistent with the literature [40,41]. A decrease in the flame size (in terms of both flame length
 439 and width) with the increase of P_{am} , can also be seen in Figs. 7(d), 8(d) and 9(d). An evaluation
 440 of the reaction zone thickness based on the full-width at half maximum of OH mass fraction at
 441 the middle of the spray flames shows that the thickness of the reaction zone is approximately 7.2,
 442 3.2 and 1.6 mm for the 14.8, 30.0 and 58.5 kg/m³ cases, respectively. These trends are in
 443 qualitative agreement with the experimental images based on OH chemiluminescence [11]. The
 444 reaction zone thickness for the O radicals exhibits a similar trend as those of OH. The spatial NO
 445 distribution becomes narrower when P_{am} increases. Although the reaction zone size decreases
 446 with the increasing ambient pressure, the rate of NO formation increases more rapidly with
 447 increasing P_{am} . The increased reaction rate attributed by the increased P_{am} is more influential to
 448 the final NO production. The total NO formation hence increases with P_{am} , as indicated by the
 449 magenta dotted lines in Fig. 12. The current model predicts that the peak soot volume fraction
 450 ($f_{v,soot}$) increases approximately six-fold when P_{am} increases from 42 to 85 bar and further
 451 increases approximately thirteen-fold when P_{am} further increases from 85 to 170 bar. To date, the
 452 revised multi-step model has only been evaluated using $f_{v,soot}$ measurements collected at 42 and
 453 85 bar [14,20] since $f_{v,soot}$ measurement above 85 bar is not available. Further work is thus
 454 required to confirm the estimation of $f_{v,soot}$ at 170 bar. Yet, the trend of maximum $f_{v,soot}$ increasing
 455 with P_{am} agrees with experimental observations [13].

456 Figure 2 shown earlier illustrates that the ignition site is placed further upstream and the
 457 flame stabilizes more quickly when P_{am} increases. At quasi-steady state, the flame LOL reduces
 458 with increasing P_{am} . This reduces the air entrainment, giving rise to a higher mixture fraction
 459 within the fuel-rich core region. Besides that, the elevated reaction rate at higher P_{am} also
 460 promotes soot formation. These factors are known to increase the local peak $f_{v,soot}$ [13]. The
 461 current study shows that the rise of the soot formation rates can be attributed to other factors. As
 462 aforementioned, the flame width reduces for a higher P_{am} . Consequently, the stoichiometric line
 463 or the mixing-controlled combustion layer is closer to the central region with higher P_{am} , which

464 increases the heat transfer. As shown in Fig. 13, the temperature in the rich premixed region
 465 increases with P_{am} , contributing to the higher soot formation rates. Apart from these, it is noticed
 466 that both the velocity near the spray region and χ decrease with the rise of P_{am} . This indicates the
 467 mixing rate within the fuel-rich region decreases, promoting soot precursor and particle formation
 468 within the fuel-rich region. These collectively contribute to the higher local $f_{v,soot}$ when P_{am}
 469 increases.



471 Fig. 13. Scalar dissipation rate, χ (solid lines) and temperature (dashed lines) along the spray axis.
 472 Semi-transparent windows indicate the domains with high soot volume fraction as can be seen in
 473 Figs. 7 to 9.

476 Concluding remarks

477 The contribution of this study is the generalization of the ignition behaviors and flame structures
 478 at different ambient pressures (P_{am}) under diesel engine-like conditions. Diesel #2 spray flames at
 479 three P_{am} levels of 42, 85 and 170 bar have been studied using an Eulerian-based transported
 480 probability density function model in three-dimensional computational fluid dynamic
 481 simulations. Comparisons to measurements show that the ignition delay times and flame lift-off
 482 lengths are predicted fairly well. The numerical results show that the mixture fraction of the first
 483 igniting mixture during the high-temperature ignition does not vary monotonically with P_{am} . The
 484 high-temperature ignition takes place on the fuel-rich side in the 42 bar and 85 bar cases.
 485 However, at 170 bar the high-temperature ignition occurs on the fuel-lean side due to the

increased reaction rates and temperature in the fuel-lean mixture. The main ignition in all three cases occurs in the region where the scalar dissipation rate (χ) is low and the residence time is long. Analyses of the quasi-steady spray flame structures reveal that the stoichiometric mixtures are closer to the fuel-rich core when P_{am} is increased, leading to higher heat transfer and increased temperature within the fuel-rich core. Besides that, χ reduces in cases with higher P_{am} . These collectively contribute to the higher local soot volume fraction when P_{am} increases, in addition to poorer air entrainment attributed by the reduced lift-off length and the increased reaction rates.

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